

Adaptive Moving Meshes in Large Eddy Simulation for Turbulent Flows

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TECHNISCHE
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joint work with C. Hertel, J. Fröhlich (TU Dresden),
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Adaptive Numerical Methods
for Partial Differential Equations with Applications
Banff, May 28th to June 1st 2018

Overview

- Short introduction to large eddy simulation (LES)
- Motivation for using moving meshes
- Physically and mathematically based r-adaptation
- Turbulent flow over periodic hills
- Baroclinically unstable jet flow
- Stationary low Mach number combustion
- Summary

Incompressible Navier-Stokes-Equation

$$\begin{aligned}\partial_t \mathbf{u} + (\mathbf{u} \cdot \nabla) \mathbf{u} + \nabla p - \nabla \cdot (2\nu \mathbf{S}(\mathbf{u})) &= \mathbf{f}, & \text{in } (0, T] \times \Omega \\ \nabla \cdot \mathbf{u} &= 0, & \text{in } (0, T] \times \Omega \\ \mathbf{u} &= \mathbf{u}_d, & \text{on } (0, T] \times \partial\Omega_D \\ p\mathbf{n} - 2\nu \mathbf{S}(\mathbf{u})\mathbf{n} &= 0, & \text{on } (0, T] \times \partial\Omega^- \\ \mathbf{u} &= \mathbf{u}_0(\mathbf{x}), & \text{in } \{0\} \times \Omega\end{aligned}$$

with $\Omega \in \mathbb{R}^3$ and $\mathbf{S}(\mathbf{u}) = (\nabla \mathbf{u} + \nabla \mathbf{u}^T)/2$.

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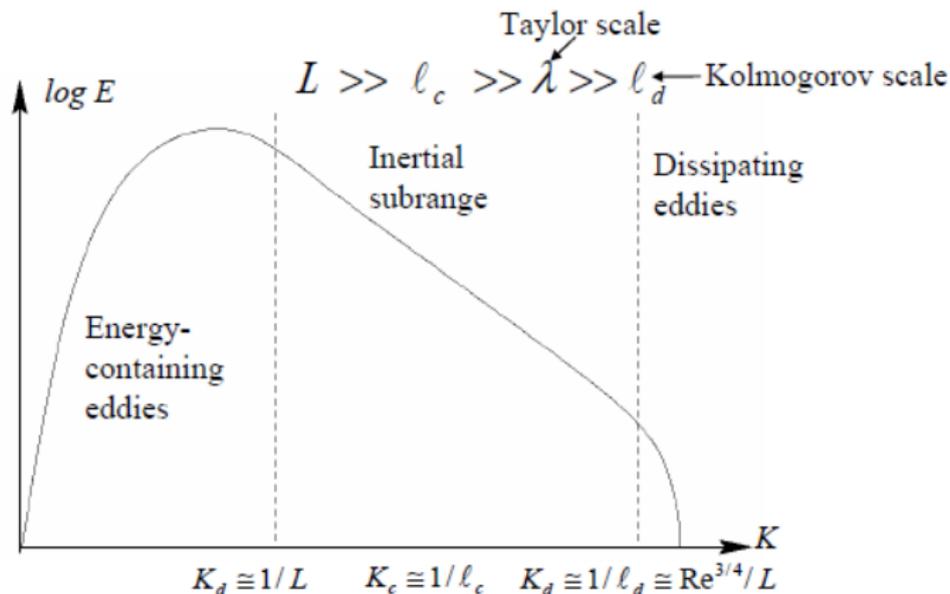
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 \end{aligned}$$

where the stress tensor $\boldsymbol{\tau}(\mathbf{u}, \bar{\mathbf{u}}) = \overline{\mathbf{u} \mathbf{u}} - \bar{\mathbf{u}} \bar{\mathbf{u}}$ has to be modelled.

Large Eddy Simulation



Goal: LES models the smallest (and most expensive) scales and resolves large scales of the flow field solution.

Miracle: Turbulent Motion of Fluids

Werner Heisenberg was asked what he would ask God, given the opportunity. His reply was:

*When I meet God, I am going to ask him two questions: Why relativity?
And why turbulence? I really believe he will have an answer for the first.*

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Horace Lamb (who had published a noted text book on Hydrodynamics) was quoted as saying

I am an old man now, and when I die and go to heaven there are two matters on which I hope for enlightenment. One is quantum electrodynamics, and the other is the turbulent motion of fluids. And about the former I am rather optimistic.

Filtered Incompressible Navier-Stokes-Equation

Use eddy viscosity model (Smagorinsky model)

$$\boldsymbol{\tau}(\mathbf{u}, \bar{\mathbf{u}}) \approx \boldsymbol{\tau}_s(\bar{\mathbf{u}}) = -\nu_t(\bar{\mathbf{u}}) \mathbf{S}(\bar{\mathbf{u}})$$

with turbulent viscosity defined by

$$\nu_t(\bar{\mathbf{u}}) = (c_s \Delta)^2 \sqrt{2} \|\mathbf{S}(\bar{\mathbf{u}})\|, \quad \|\mathbf{S}(\bar{\mathbf{u}})\| = (\mathbf{S}(\bar{\mathbf{u}}) : \mathbf{S}(\bar{\mathbf{u}}))^{1/2}$$

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Closed model for $(\bar{\mathbf{u}}_s, \bar{p}_s)$:

$$\begin{aligned} \partial_t \bar{\mathbf{u}}_s + (\bar{\mathbf{u}}_s \cdot \nabla) \bar{\mathbf{u}}_s + \nabla \bar{p}_s - \nabla \cdot ((2\nu + \nu_t) S(\bar{\mathbf{u}}_s)) &= \bar{\mathbf{f}}, & (0, T] \times \Omega \\ \nabla \cdot \bar{\mathbf{u}}_s &= 0, & (0, T] \times \Omega \\ \bar{\mathbf{u}}_s &= \bar{\mathbf{u}}_d, & (0, T] \times \partial\Omega_D \\ \bar{p}_s \mathbf{n} - (2\nu + \nu_t) S(\bar{\mathbf{u}}_s) \mathbf{n} &= 0, & (0, T] \times \partial\Omega^- \\ \bar{\mathbf{u}}_s &= \bar{\mathbf{u}}_0(\mathbf{x}), & \{0\} \times \Omega \end{aligned}$$

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Good news: Nowadays LES works quite well provided

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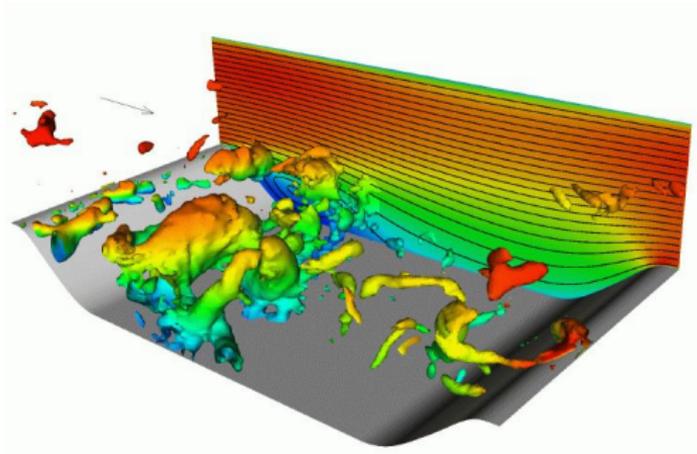
Need for a posteriori quality improvement of LES!

LES - Statistics

- LES produces huge data of space- and time-resolved flow solutions, but often one is only interested in **statistical values** as mean velocities and fluctuations, which can be compared to experimental data.

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LES for flow over periodic hills, $Re = 10595$.

LES - Statistics

- Define **time-averaging**

$$\langle \mathbf{v} \rangle(\mathbf{x}) = \lim_{T \rightarrow \infty} \frac{1}{T} \int_0^T \mathbf{v}(t, \mathbf{x}) dt$$

and **fluctuations**

$$\mathbf{v}'' = \mathbf{v} - \langle \mathbf{v} \rangle$$

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- For LES it holds

$$\langle \bar{\mathbf{u}}'' \bar{\mathbf{u}}'' \rangle \approx \langle \tau(\mathbf{u}, \bar{\mathbf{u}}) \rangle + \langle \bar{\mathbf{u}} \bar{\mathbf{u}} \rangle - \langle \bar{\mathbf{u}} \rangle \langle \bar{\mathbf{u}} \rangle,$$

which gives a possibility to approximate the time-averaged stress tensor $\langle \tau(\mathbf{u}, \bar{\mathbf{u}}) \rangle$.

Motivation for Mesh Adaptation

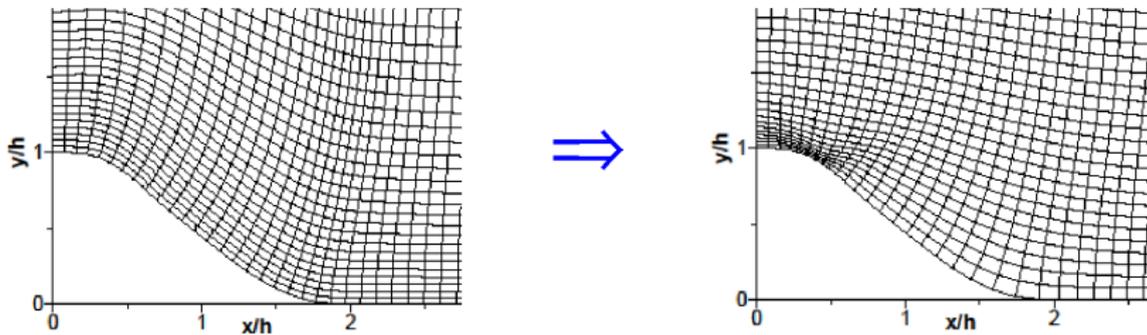
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- Appropriate monitor functions for LES
- **Adaptive scale separation** w.r.t. **time-averaged solution**



Mesh Moving Method

- Physical domain Ω with coordinates $\mathbf{x} = (x_1, x_2, x_3)^T$
- Computational domain Ω_c with coordinates $\boldsymbol{\xi} = (\xi_1, \xi_2, \xi_3)^T$

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- Minimise mesh adaptation functional (equidistribution principle)

$$\mathcal{I}[\boldsymbol{\xi}] = \frac{1}{2} \int_{\Omega} \sqrt{g} \sum_{i=1}^3 \nabla \xi_i G^{-1} \nabla \xi_i d\mathbf{x}, \quad g = \det(G)$$

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- Derive time-dependent mesh moving PDE

Mesh Moving Method

Time-dependent **Mesh Moving PDE** [Huang, Russell]:

$$\tau \frac{\partial \mathbf{x}}{\partial t} = \frac{1}{P} \left(\sum_{i,j} a_{ij} \frac{\partial^2 \mathbf{x}}{\partial \xi_i \partial \xi_j} - \sum_i b_i \frac{\partial \mathbf{x}}{\partial \xi_i} \right)$$

where

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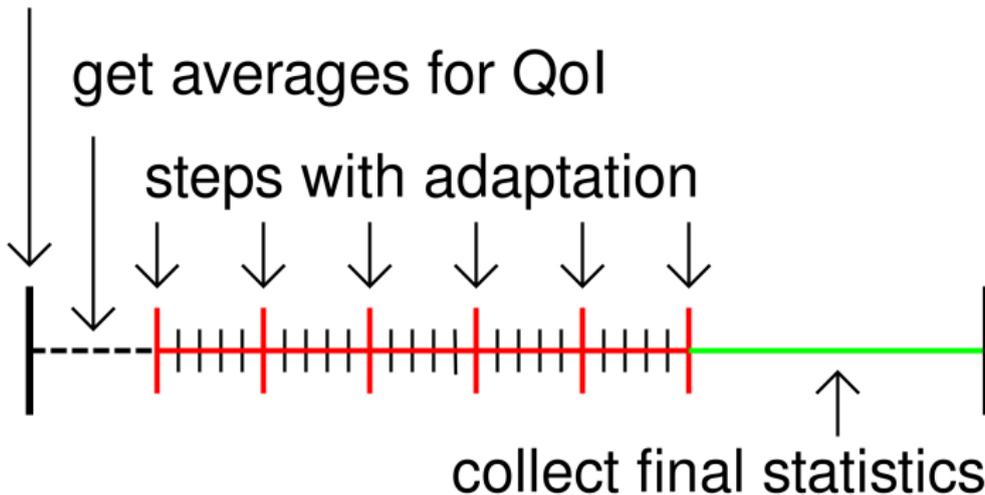
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Monitor function should depend on some **quantity of interest Ψ** , physically or mathematically motivated.

LES with Moving Meshes: General Strategy

statistically converged flow on stationary grid



Mesh Moving Method - Implementation

- Use [LESOCC2](#) - advanced parallel code for engineering applications
- Second order [cell-centered finite volume method](#) for curvilinear coordinates, coupled with predictor-corrector scheme based on three-stage Runge-Kutta methods and pressure correction equation

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- Implement arbitrary Lagrangian-Eulerian formulation (ALE) with time-varying control volumes $V(t)$ and surfaces $S(t)$
- Ensure **mass conservation** via **space conservation law** [Demirdzic, Peric]

$$\frac{d}{dt} \int_{V(t)} dv - \int_{S(t)} u_N \mathbf{n} ds = 0$$

where u_N is the (given) vector of node velocities. Adapt mesh movement.

Mesh Moving Method - Implementation

Cell Centered Finite Volume Method

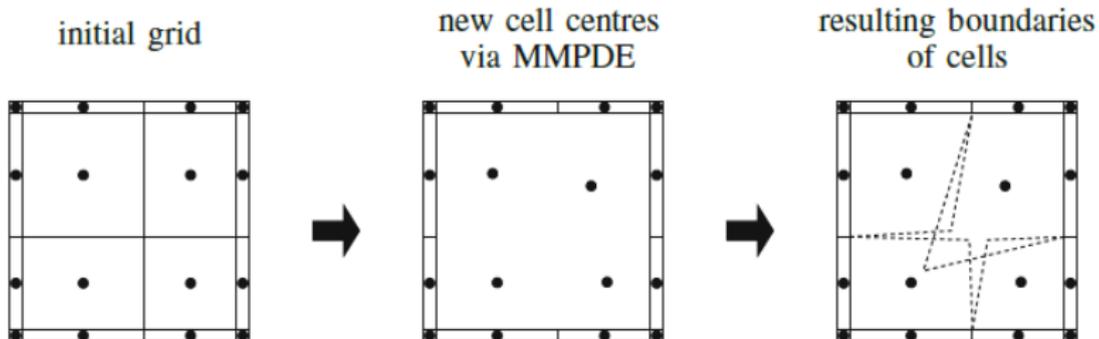


Illustration of difficulties when constructing a valid grid from cell centres

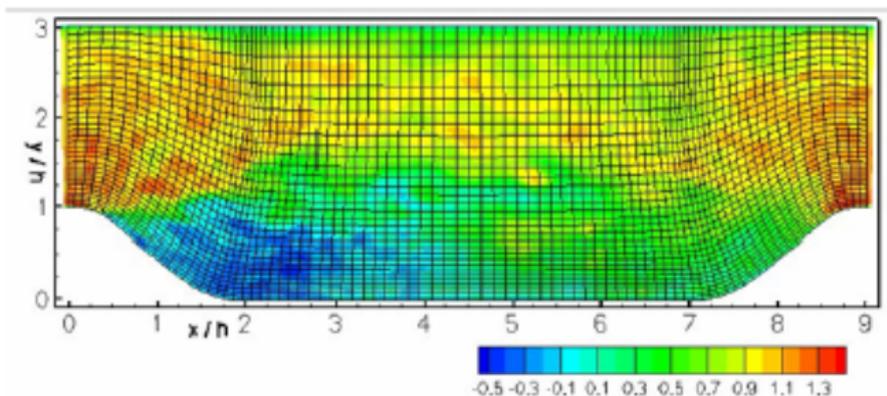
Mesh Moving Method - Implementation

Cell Centered Finite Volume Method

- Step 1 Integration of the MMPDE for cell centres with fixed points at boundaries yielding preliminary values of cell centres \mathbf{x}^* .
- Step 2 Determination of cell corner points $\tilde{\mathbf{x}}^{n+1}$ via an interpolation method.
- Step 3 Determination of corner points on boundaries from final corner grid in the interior.
- Step 4 Re-computation of cell centres \mathbf{x}^{n+1} in the domain and on the boundary to generate the final valid grid.

Hertel, Schümichen, JL, Fröhlich: *Using a Moving Mesh PDE for Cell Centres to Adapt a Finite Volume Grid*, Flow, Turbulence and Combustion 90(4), pp. 785–812, 2013.

Turbulent Flows over Periodic Hills



- $Re = 10595$
- Smagorinsky subgrid-scale model with $c_s = 0.1$
- Reference solution with 4.500.000 cells [Fröhlich et al., 2005]
- Computational grid: $89 \times 33 \times 49$ (135.168 cells)

The Engineer's Approach

Quantity of Interest Φ

Here: $\langle \cdot \rangle$ denotes averaging in homogeneous direction and time.

- Gradient of streamwise velocity: $\Phi = \nabla \langle \bar{u}_1 \rangle$
- Modelled turbulent kinetic energy (TKE)

$$\Phi = \frac{\langle k_{sgs} \rangle}{k_{tot,max}}, \quad k_{sgs} \approx (2^{1/3} - 1)0.5|\bar{\mathbf{u}} - \bar{\bar{\mathbf{u}}}| \quad [\text{Berselli et al., 2006}]$$

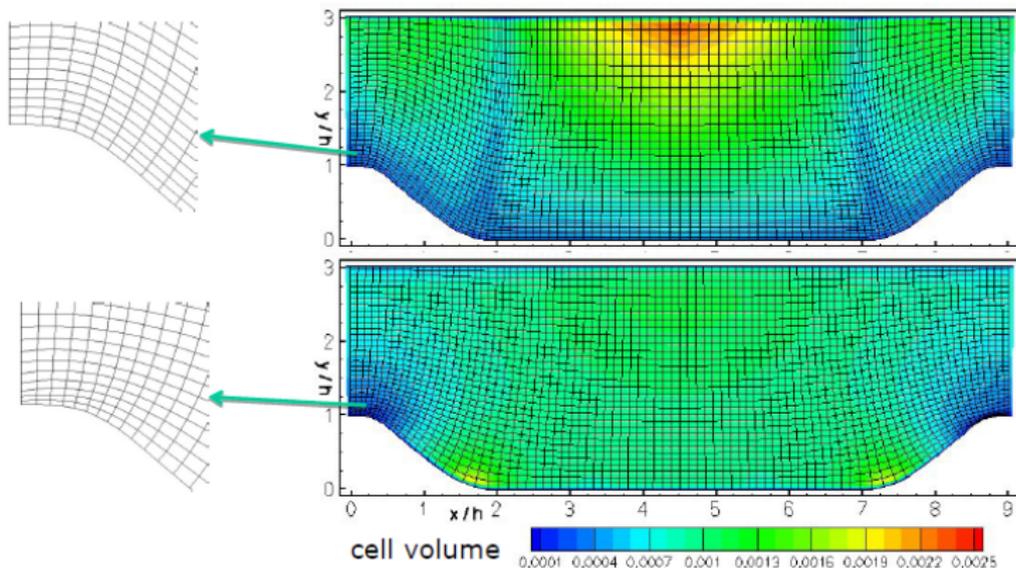
- Turbulent shear stress: ratio of modelled and total shear stress

$$\Phi = \frac{\langle \tau_{12}^{mod} \rangle}{\langle \tau_{12}^{mod} \rangle + \langle \bar{u}_1'' \bar{u}_2'' \rangle}, \quad \tau_{12}^{mod} = -\nu_t \left(\frac{\partial \bar{u}_1}{\partial x_2} + \frac{\partial \bar{u}_2}{\partial x_1} \right)$$

Use combination of them as well.

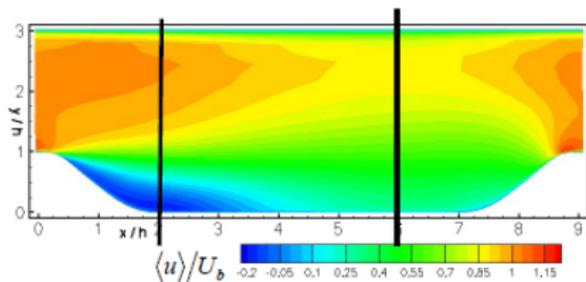
The Engineer's Approach

Monitor function: gradient of streamwise velocity $\Phi = \nabla \langle \bar{u}_1 \rangle$

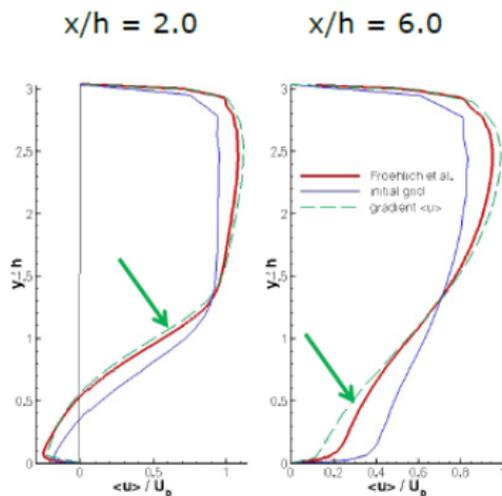


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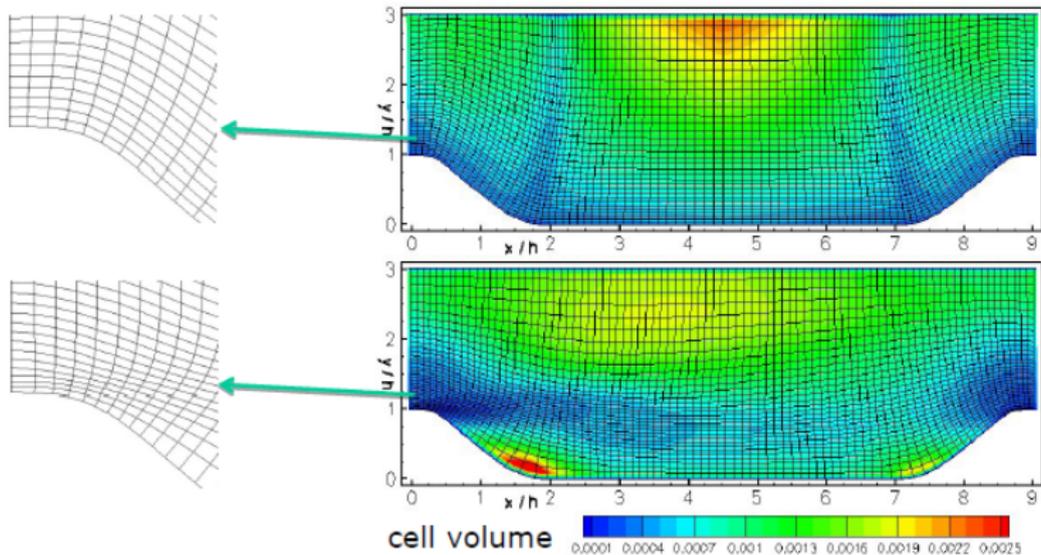
	x_{sep}/h	x_{rea}/h
Reference	0.2	4.6
Initial grid	0.5	3.1
with adaptation	0.3	4.7



Comparison of low separation and reattachment point

The Engineer's Approach

Monitor function: combine gradient of streamwise velocity and modelled kinetic energy $\Phi = \nabla \langle \bar{u}_1 \rangle + \frac{\langle k_{sgs} \rangle}{k_{tot,max}}$



The Engineer's Approach

Comparison of low separation and reattachment point

	x_{sep} / h	x_{rea} / h
Reference (5 Mio. cells)	0.2	4.6
Initial grid	0.5	3.1
1. $\nabla \langle u \rangle$	0.3	4.7
2. $\frac{\langle k_{zgr} \rangle}{k_{\text{tor,max}}}$	0.45	3.4
3. $\frac{\langle \tau_{12}^{\text{mod}} \rangle}{\langle \tau_{12}^{\text{mod}} \rangle + \langle u'v' \rangle}$	0.45	4.15
4. $\psi_1 = \nabla \langle u \rangle$ & $\psi_2 = \frac{\langle k_{zgr} \rangle}{k_{\text{tor,max}}}$	0.3	4.5
5. $\psi_1 = \nabla \langle u \rangle$ & $\psi_2 = \frac{\langle \tau_{12}^{\text{mod}} \rangle}{\langle \tau_{12}^{\text{mod}} \rangle + \langle u'v' \rangle}$	0.25	4.55

← best

The Mathematician's Approach

Dual Weighted Residual Method [Becker, Rannacher, Braack, ...]

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Goal: quantify the contributions of the subgrid-scale model and the numerical method to a user specified quantity of interest

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Linearize N in the neighbourhood of $(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle)$:

$$M(\langle \mathbf{u} \rangle, \langle p \rangle) - M(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle) = \int_{\Omega} \left\{ \partial_{\langle \mathbf{u} \rangle} N(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle) \mathbf{e}_{\langle \mathbf{u} \rangle} + \partial_{\langle p \rangle} N(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle) e_{\langle p \rangle} \right\} dx + H.O.T.$$

with $\mathbf{e}_{\langle \mathbf{u} \rangle} = \langle \mathbf{u} \rangle - \langle \bar{\mathbf{u}}_h \rangle$ and $e_{\langle p \rangle} = \langle p \rangle - \langle \bar{p}_h \rangle$.

The Mathematician's Approach

Define (linear) **stationary dual system** with (ϕ, θ) :

$$-\langle \bar{\mathbf{u}}_h \rangle \cdot \nabla \varphi + (\nabla \langle \bar{\mathbf{u}}_h \rangle)^T \varphi + \nabla \theta$$

$$-\nabla \cdot ((2\nu + \nu_t(\langle \bar{\mathbf{u}}_h \rangle))S(\varphi)) - \nabla \cdot T^h[\langle \bar{\mathbf{u}}_h \rangle](\varphi) = \partial_{\langle \mathbf{u} \rangle} N(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{\mathbf{p}}_h \rangle)$$

$$-\nabla \cdot \varphi = \partial_{\langle \mathbf{p} \rangle} N(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{\mathbf{p}}_h \rangle)$$

$$\varphi = 0 \quad \text{b.c.}$$

$$\varphi(T, \mathbf{x}) = 0 \quad \text{i.c.}$$

where $T^h[\langle \bar{\mathbf{u}}_h \rangle](\varphi) = (c_s \Delta)^2 \|S(\langle \bar{\mathbf{u}}_h \rangle)\|_F^{-1} (S(\langle \bar{\mathbf{u}}_h \rangle) : S(\varphi))S(\langle \bar{\mathbf{u}}_h \rangle)$.

The Mathematician's Approach

Theorem [Computable error representation formula]

Let $(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle)$ be the numerical solution and N a given operator. Then

$$M(\langle \mathbf{u} \rangle, \langle p \rangle) - M(\langle \bar{\mathbf{u}}_h \rangle, \langle \bar{p}_h \rangle) \approx e_M + e_N + H.O.T.$$

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with e_M and e_N are given by

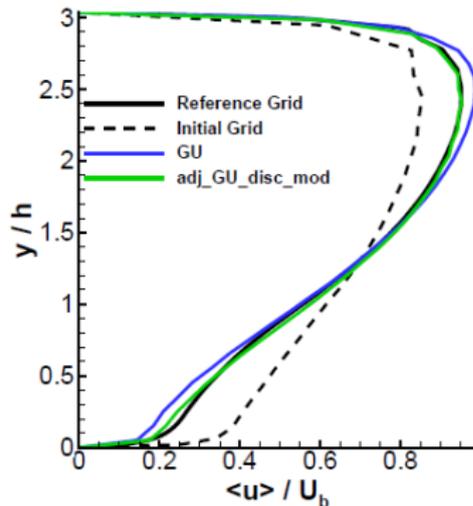
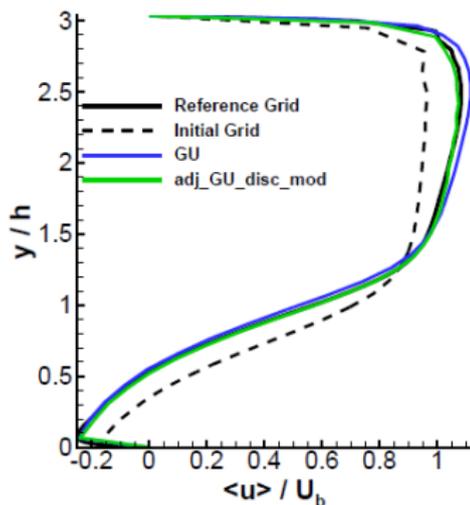
$$e_M = \int_{\Omega} \phi_h \{ \langle \mathbf{f} \rangle - \langle \bar{\mathbf{f}} \rangle + \langle (\nabla \cdot \tau_s)(\bar{\mathbf{u}}_h) \rangle - \langle \nabla \cdot \tau_{ds}(\bar{\mathbf{u}}_h) \rangle \} d\mathbf{x}$$

$$e_N = \int_{\Omega} \{ \phi_h \langle ResSM(\bar{\mathbf{u}}_h, \bar{p}_h) \rangle + \theta_h \langle \nabla \cdot \bar{\mathbf{u}}_h \rangle \} d\mathbf{x}$$

where $ResSM(\bar{\mathbf{u}}_h, \bar{p}_h)$ is the residual of the space-averaged momentum equation with Smagorinsky subgrid-scale model.

The Mathematician's Approach

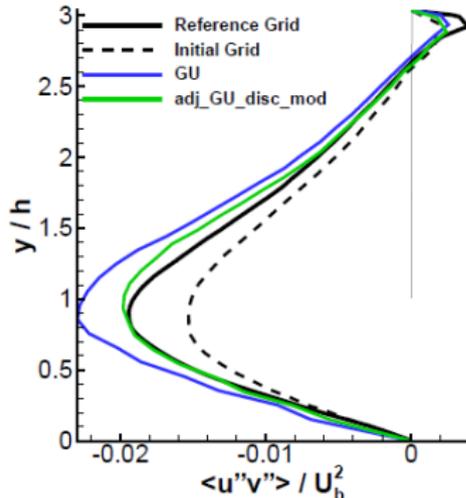
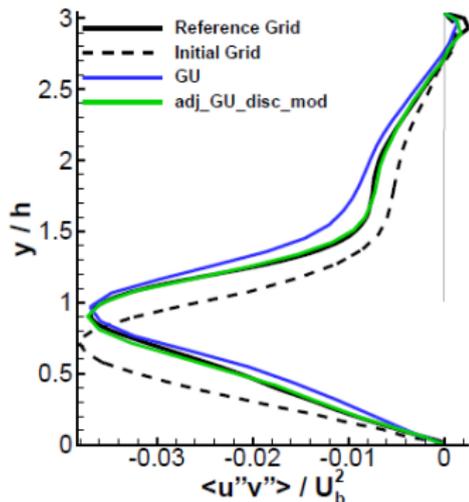
Monitor function: $\Phi = \Phi_N + \Phi_M$ based on $N(\bar{\mathbf{u}}) = \nabla \langle \bar{\mathbf{u}}_1 \rangle$



Comparison of $\nabla \langle \bar{\mathbf{u}}_1 \rangle / U_b$ at $x/h = 2$ and $x/h = 6$.

The Mathematician's Approach

Monitor function: $\Phi = \Phi_N + \Phi_M$ based on $N(\bar{\mathbf{u}}) = \nabla \langle \bar{\mathbf{u}}_1 \rangle$



Comparison of $\langle \bar{\mathbf{u}}_1'' \bar{\mathbf{u}}_2'' \rangle / U_b^2$ at $x/h = 2$ and $x/h = 6$.

Meteorological Application

German Priority Program 'METSTROEM' (Funded by DFG)

Collaborators in the first period: C. Kühnlein, A. Dörnbrack (Munich),
P.K. Smolarkiewicz (Boulder)

Software Package MPDATA and EULAG

Meteorological Application

German Priority Program 'METSTROEM' (Funded by DFG)

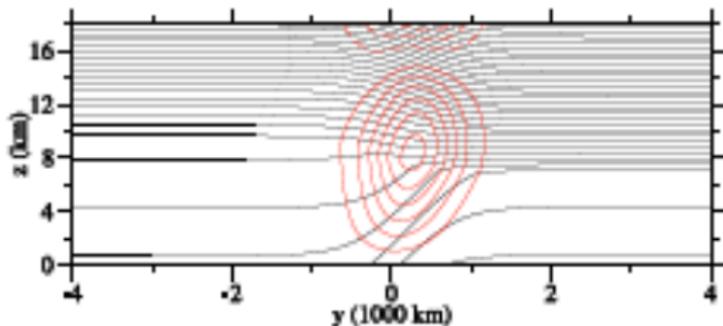
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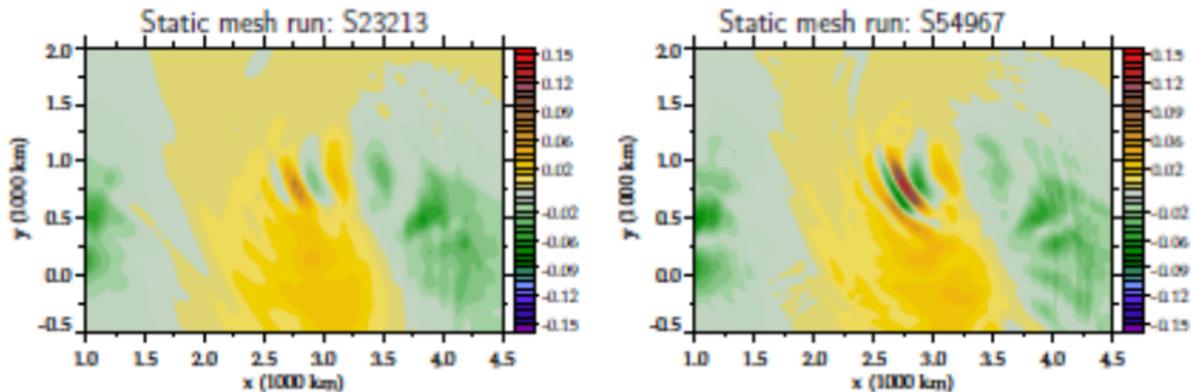
Application:

Baroclinically unstable jet flow in inviscid and dry atmosphere

Zonally-periodic channel: 10.000 km \times 8.000 km \times 18 km



Meteorological Application: Baroclinically Unstable Jet Flow in Inviscid and Dry Atmosphere

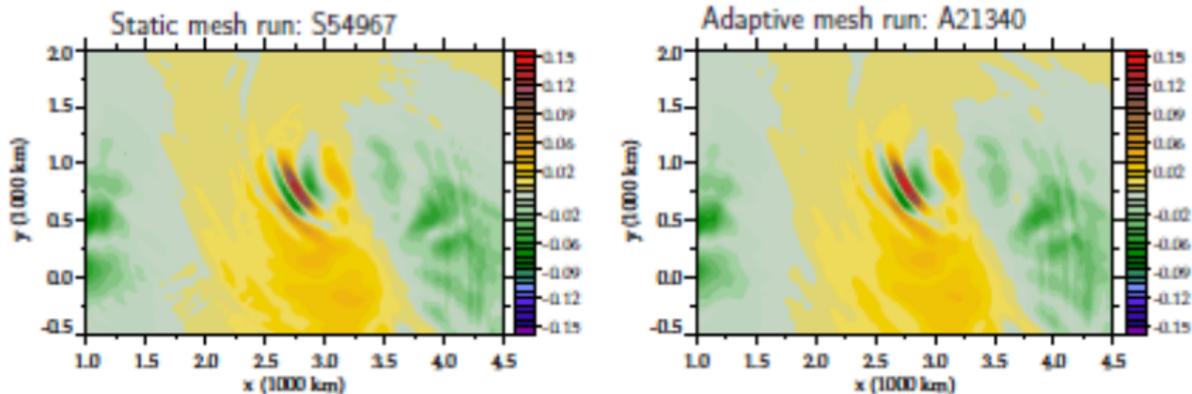


Representation of mesoscale internal gravity waves

$$\Phi = 1/H \int_0^H \|\nabla\Theta(t, x, y, z)\| dz$$

(C. Kühnlein, A. Dörnbrack, P.K. Smolarkiewicz, 2011)

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Low Mach Number Compressible Combustion

$$\rho \partial_t \mathbf{u} + \rho (\mathbf{u} \cdot \nabla) \mathbf{u} + \nabla p_{hyd} - \nabla \cdot (2\nu \mathbf{S}(\mathbf{u})) = \rho \mathbf{g}, \quad \text{in } (0, T] \times \Omega$$

$$\frac{1}{M} \mathbf{u} \cdot \nabla M - \frac{1}{T} \mathbf{u} \cdot \nabla T + \nabla \cdot \mathbf{u} = 0, \quad \text{in } (0, T] \times \Omega$$

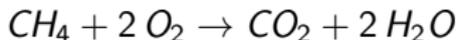
$$c_p \partial_t T + c_p \rho \mathbf{u} \cdot \nabla T - \nabla \cdot (\lambda \nabla T) = f_T(T, \boldsymbol{\omega}), \quad \text{in } (0, T] \times \Omega$$

$$\partial_t \omega_i + \rho \mathbf{u} \cdot \nabla \omega_i - \nabla \cdot (\rho D_i \nabla \omega_i) = f_i(T, \boldsymbol{\omega}), \quad \text{in } (0, T] \times \Omega$$

$$i = 1 : N, \quad \Omega \subset \mathbb{R}^2$$

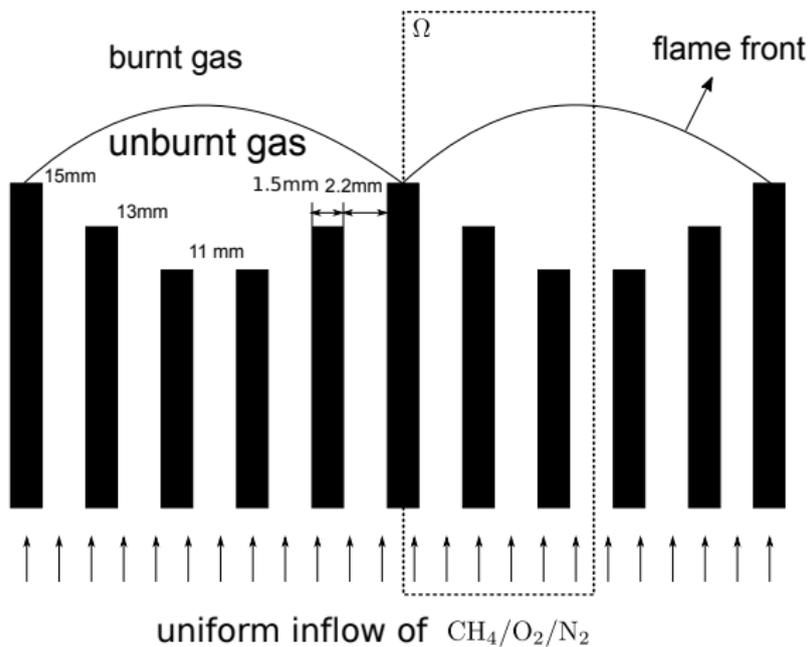
$$\rho = \frac{P_{th} M}{RT}, \quad \frac{1}{M} = \sum_{i=1}^N \frac{\omega_i}{M_i}$$

Methane Burner (JUNKERS Bosch Thermotechnik, 1988) with global reaction



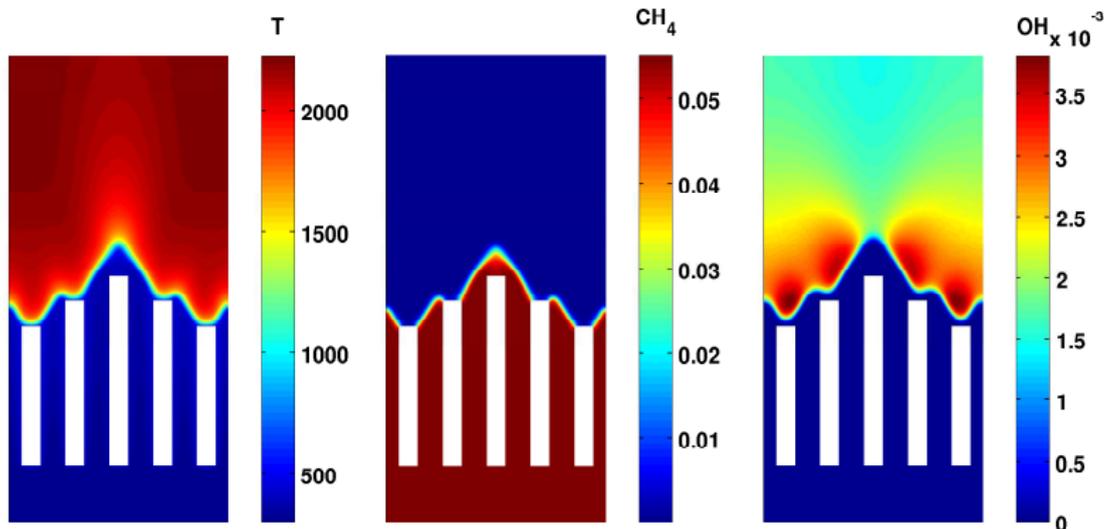
$N = 15$ species and 84 elementary reactions (no NO_x formation)

Methane Burner



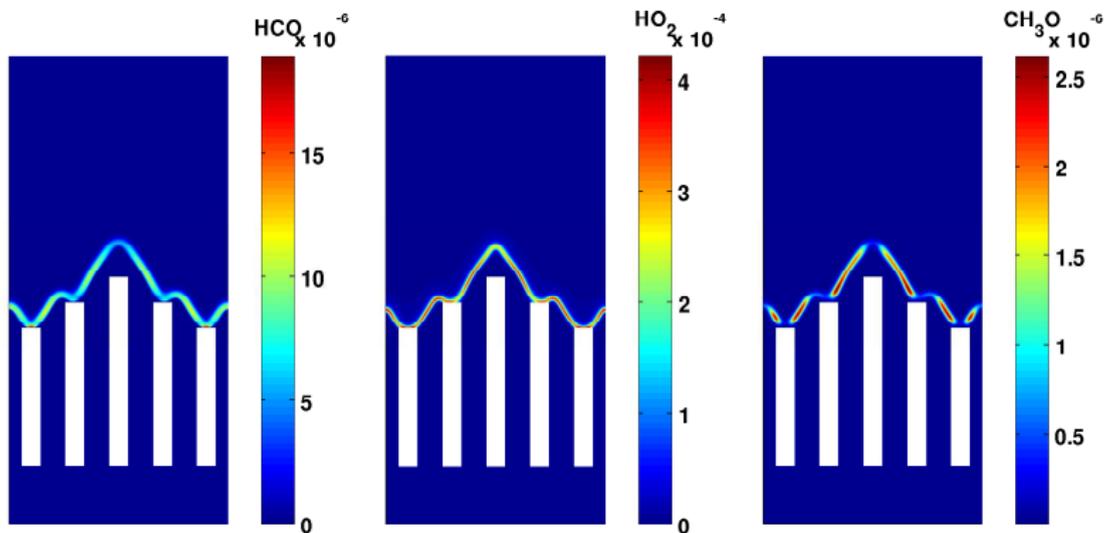
Geometry of the Junkers Bosch Methane Burner

Methane Burner



Stationary solutions: T, CH_4 , and OH

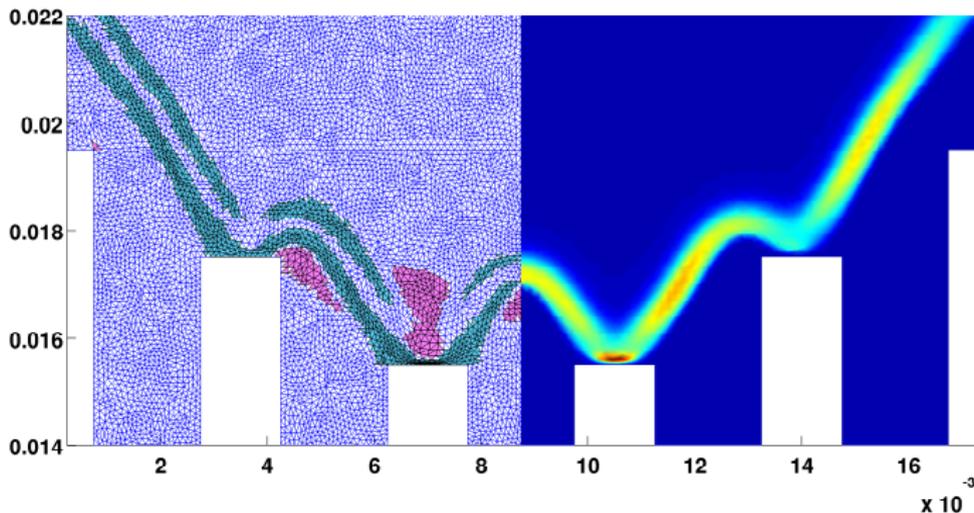
Methane Burner



Stationary solutions: HCO (Hydrocarbonate), HO_2 , and CH_3O

Moving Meshes for the Methane Burner

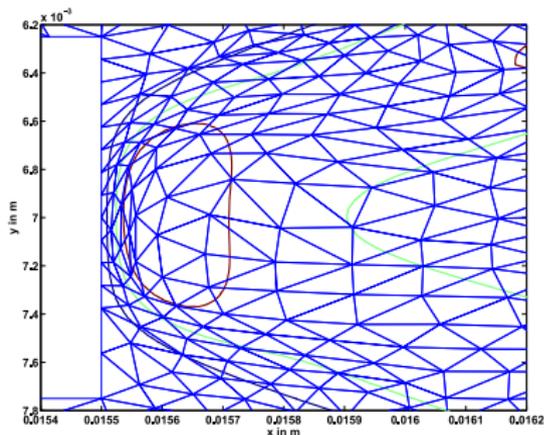
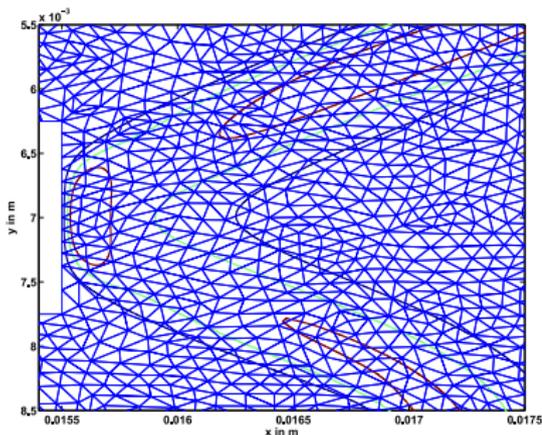
Quantity of Interest: $\Phi = \nabla \omega_{CHO}$



Profile of radical HCO. The elements in magenta represent enlarged triangles (rate: +0.15), and the cyan ones show the compressed cells (rate: -0.05).

Moving Meshes for the Methane Burner

Quantity of Interest: $\Phi = \nabla \omega_{CHO}$



Initial mesh (left) and adaptive mesh (right) close to the shortest slot.

Summary

- LES is well suited for moving mesh techniques.
- Physically motivated monitor functions work quite well for LES.
- High potential of sensitivity-based mesh moving methods based on adaptive scale separation.
- Resolve as much physics as possible with given DoFs: Model dissipation and resolve production.
- Application of moving meshes to complex combustion still needs expert knowledge.

Final Remark

Independent (UK), 4th May 2018
Karl Marx 200th anniversary

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